

Review

# The Damage identification based guided waves in composite structures with AI schemes: A review

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**Abstract:** Lamb waves (LWs) are guided waves (GWs) with unique properties that make them ideal for Non-Destructive Testing (NDT) and Structural Health Monitoring (SHM). Over the last two decades, LW-based techniques have gained significant traction in academic and industrial research. A primary advantage of LWs is their ability to propagate over long distances in plate or shell structures, navigating curves and reaching hidden or buried areas, capabilities essential for detecting interlaminar damage in composites. As LW wave structures are frequency and phase-velocity-dependent, they are now suitable for various engineering applications, offering a promising, cost-effective solution for automated, real-time structural integrity monitoring. This paper provides a comprehensive overview of LW behavior, modeling, applications, and limitations in detecting fatigue damage within composite structures. Furthermore, the study explores the integration of Artificial Intelligence (AI) algorithms, including Artificial Neural Networks (ANNs), Machine Learning (ML), and Deep Learning (DL), which have emerged as powerful tools for predicting fatigue damage. By leveraging historical data from LW sensing systems, these AI models can accurately forecast structural conditions without the need for complex analytical modeling, significantly reducing processing time.

**Keywords:** Lamb waves technique; piezoelectric transducers (PZTs); fatigue damage detection; composite structures; artificial neural networks (ANNs); machine learning (ML); deep learning (DL)

## 1. Introduction

Composite materials have become widely used in engineering across a variety of fields. It's particularly important in industries like aviation, aerospace, and automotive, and is also utilized in constructing ships, submarines, and nuclear and chemical facilities. These materials offer impressive stiffness and strength relative to their weight, as well as enhanced resistance to fatigue and damage.

The expanding use of composite materials in increasingly ambitious projects, such as taller buildings, longer bridges, and larger aircraft, is driving a critical need for advanced structural health monitoring and damage detection.

Common issues that can arise include things like fiber breakage, matrix cracking, debonding between fibers and the matrix, and delaminations or interlayer cracks. These types of damage can seriously impact the strength and overall structural integrity. So, catching these problems early is crucial to avoid catastrophic failures, especially since these structures are often expected to perform well near their limits.

A solid monitoring system can keep tabs on the development of structural damage, allowing for timely maintenance or repairs to maintain safety and integrity.

One effective way to achieve this is through SHM systems. By adding extra safety measures, SHM systems help maximize the lifespan of structures and can also lower the overall costs related to maintaining them.

SHM is an expanding field with applications across rotating machinery, offshore platforms, aerospace, and civil infrastructure [1,2]. Implementing SHM is critical for assets where failure poses a risk to human life, particularly in composite structures. Early detection of damage allows for timely intervention, such as restricting usage or performing essential repairs, to prevent catastrophic failure.

SHM aims to assess a structure's health by using a set of sensors that are permanently installed on it, keeping tabs over time. It starts with a check-up on the structure's safety and general condition, then looks at how much life it has left. There are two ways to do SHM: passively and actively. With passive SHM, the sensors just 'listen' to the structure to catch any changes in the components. On the other hand, active SHM involves a network of sensors that actively probe the structure's health, helping to figure out if there's any damage or not. In recent years, a variety of quantitative methods have been researched and put into practice for detecting damage in both metal and composite structures. Most of these methods depend on non-destructive (ND) techniques.

LWs techniques offer significant potential for damage detection in composite structures across diverse engineering sectors. Compared to traditional NDT methods, LW-based approaches provide faster, more cost-effective assessments of various damage types. While monitoring complex geometries requires specialized expertise, a primary advantage of LWs is the ability to inspect large areas using a minimal number of transducers. However, optimizing transducer placement remains a significant challenge. The integration of piezoelectric transducers (PZTs) with guided LWs has gained considerable traction due to their superior weight, cost, and functional benefits [3,4].

The piezoelectric effect can be thought of as a connection between mechanical and electrical aspects of a material, like mechanical stress, strain, and electric field or charge. Basically, when mechanical stress is put on a piezoelectric material, it generates an electrical charge, which we refer to as piezoelectricity. This means that the one-dimensional movement can be used to deform the piezoelectric in various ways (just look at **Figure 1**). Plus, this process works both ways; an electric field or charge can be applied to create mechanical strain or stress. **Figure 2** illustrates both the direct and reverse mechanisms involved in the piezoelectric sensing process, which is useful for monitoring structures.

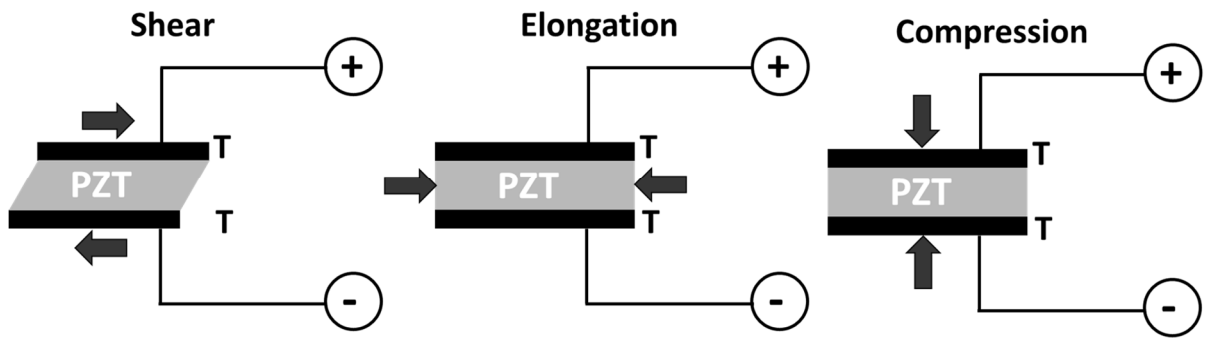


Figure 1. PZTs deformation modes.

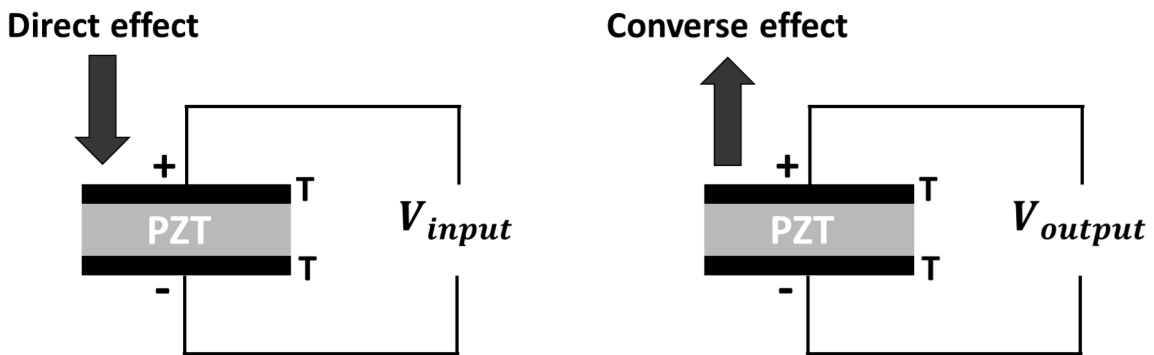


Figure 2. The direct and inverse response principle in the PZTs effect.

LWs inspection relies on waves moving through flat structures, and these waves react to any damage present. To detect changes in the LWs responses, first need to collect baseline signals that show what it looks like when there's no damage.

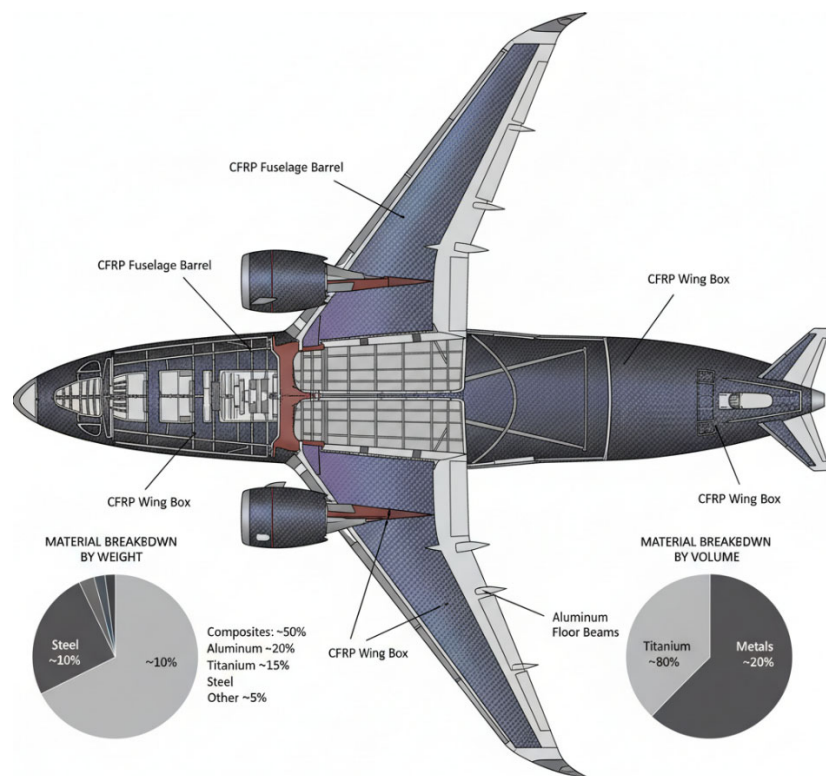
LWs offer several advantages for NDT. They can propagate over significant distances, even within high-attenuation materials like composites. Furthermore, LWs are highly sensitive to internal interference, enabling the efficient inspection of large areas such as aircraft wing skins [5]. Unlike surface-level methods, LWs can detect both surface and internal defects by utilizing various modes to access the material's full thickness. In summary, LW-based detection provides five key benefits: (1) the ability to inspect large structures through coatings or insulation; (2) full cross-sectional coverage over long distances; (3) the elimination of expensive structural probing; (4) the capacity to identify multiple defects simultaneously; and (5) low energy consumption and operational costs [5]. By analyzing LW data, engineers can diagnose, locate, and characterize the severity of fatigue damage in fiber-reinforced polymer (FRP) structures, ultimately providing more accurate estimates of their remaining useful life.

Next, looked into the various LWs methods that have been developed for detecting damage in composite structures over the years. Also, explored key topics like where to best place sensors, telemetry approaches, system identification techniques, and the use of finite element (FE) methods. One of the highlights is the use of ANNs, which have become a go-to method for predicting fatigue damage in composite structures recently.

Before diving into how fatigue damage detection works in composite materials with LWs, there are a few important points to consider. First, one needs to have a solid grasp of the composite materials themselves, including their failure mechanisms and fatigue behaviors, as well as what influences them. Then, should take a close look at the methods used to detect fatigue damage in these structures, particularly NDT techniques. Finally, it's crucial to review previous research on the topic to establish a basis for comparison.

### 1.1. Composite materials

The adoption of composite materials has increased significantly in recent years, driven primarily by their superior specific strength, stiffness, fatigue resistance, and damage tolerance. These advantageous properties have led to their widespread integration across the construction, automotive, aerospace, marine, and sporting goods industries. In aerospace specifically, composites are utilized for critical components such as radomes, wing panels, and stabilizers. A primary example of this shift is the Boeing 787 Dreamliner, which features an airframe composed largely of composite materials, as illustrated in **Figure 3** [6].



**Figure 3.** The composite materials percentages in Boeing 787.

When it comes to composite materials, key mechanical traits like strength, stiffness, and fatigue life are crucial. Figuring out the integrity and reliability of these structures is a must. Over the past few decades, composite structures have posed significant research challenges, mainly because they have been used more and more across different applications, from mechanical setups to structural systems. These

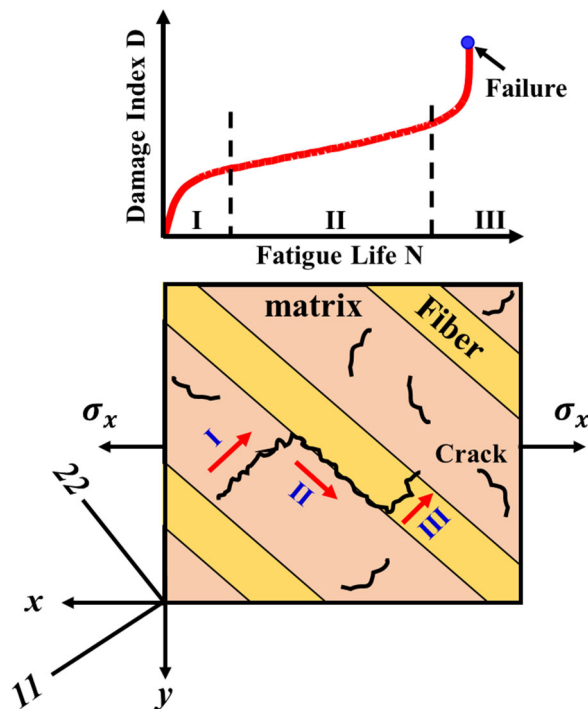
composites often face both static and fatigue loads. In fact, it's been noted that around 80% of failures in mechanical and structural components relate to fatigue [7].

### 1.2. The mechanisms of failure

Because composite materials are anisotropic, they can experience various failure modes either at the same time or one after another before the entire structure gives way.

Typically, when composites face static loading, they fail due to a mix of different interacting mechanisms that ultimately lead to a break. For laminates, as well as single layers, different types of damage mechanisms can be observed. Often, the failure happens at the point where the matrix meets the reinforcement, which is known as debonding, especially at defects that are almost always present in composites, largely because of how it's made. Other frequent failure modes include matrix cracking, fiber rupture, delamination (in laminates), and buckling (especially under compression) [8].

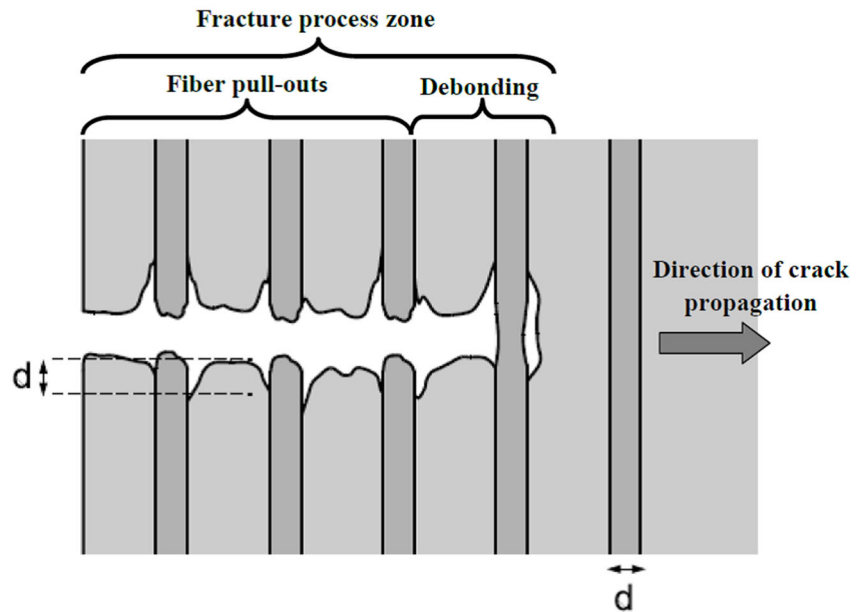
Fatigue damage in composite materials initiates early in the loading cycle. The fundamental mechanisms governing fatigue behavior under both uniaxial [9–12] and multiaxial [13–15] loading are well-documented and consistent. This damage typically progresses through three distinct stages: matrix cracking (Phase I), the initiation of local delaminations from those cracks (Phase II), and the coalescence of delaminations leading to ultimate failure (Phase III). Correspondingly, the structural stiffness exhibits three similar phases: a rapid initial degradation during the first 10–20% of the fatigue life due to matrix cracking, followed by a gradual, near-linear decline. In the final stage, damage accumulates exponentially, leading to a 'sudden death' failure [16], as illustrated in **Figure 4** [17].



**Figure 4.** The Failure mechanisms phases.

Schulte et al. [18,19] were the first to mention this three-phase stiffness reduction, and since then, it's been seen in a variety of composite materials, including woven composites [20,21].

Fiber-reinforced composites are made up of three different phases, and because of that, failure can happen in any one of those phases or at the boundaries where they meet. So, the types of failures in these composites include fiber breaks, cracks in the matrix, debonding, and delamination (**Figure 5**).



**Figure 5.** Failure modes on a laminate.

### 1.3. The fatigue behavior

The fatigue behavior of composite materials is inherently complex due to their anisotropic and heterogeneous nature. Despite extensive research into these characteristics, design methodologies often continue to rely on exhaustive fatigue testing and conservative safety factors. For unidirectional (UD) composites under axial tension-tension loading, fatigue resistance is governed by the interaction between the fibers, matrix, and their respective interfaces. These damage mechanisms are typically characterized using a Fatigue Life Diagram (FLD), which partitions fatigue behavior into three distinct regions. The specific morphology of these regions on the FLD depends on the constituent properties and interfacial bonding, as illustrated in the schematic representation in **Figure 6**.

Within the FLD, Region I represents the quasi-static failure scatter band, which remains independent of cyclic loading and is not solely governed by the matrix. In contrast, Regions II and III are primarily influenced by the fatigue properties of the matrix, though they are also affected by fiber stiffness, interfacial bonding, matrix inelasticity, and the specific loading conditions. Specifically, in Region II, stiffer fibers can impede the growth of fiber-bridged cyclic cracks, thereby extending fatigue life. Furthermore, high-stiffness fibers more effectively arrest matrix fatigue cracks, increasing the fatigue limit, a trend indicated by the arrows in **Figure 6**. A comparison between glass-epoxy (GFRE) and carbon-epoxy (CFRE) composites in **Figure 7**

clearly demonstrates this effect; the transition from glass to carbon fibers results in a significant shift of Region II toward a longer fatigue life.

The two regions, Regions II and III, in FLD are significantly influenced by some other critical factors, and these can be ranked based on their impact on FLD: (1) loading types [22,23], (2) loading frequency [24–26], (3) volume fraction ( $V_f$ ) [27–29], (4) fiber orientation [30,31], (5) mean stress and stress ratio [32,33], (6) environmental factors [34,35], (7) sizing and stress gradient [36,37], (8) surface finish [38,39], and (9) stress concentration [36,40] (9) adhesive joint type [41,42], (10) thermal and other factors [43,44].

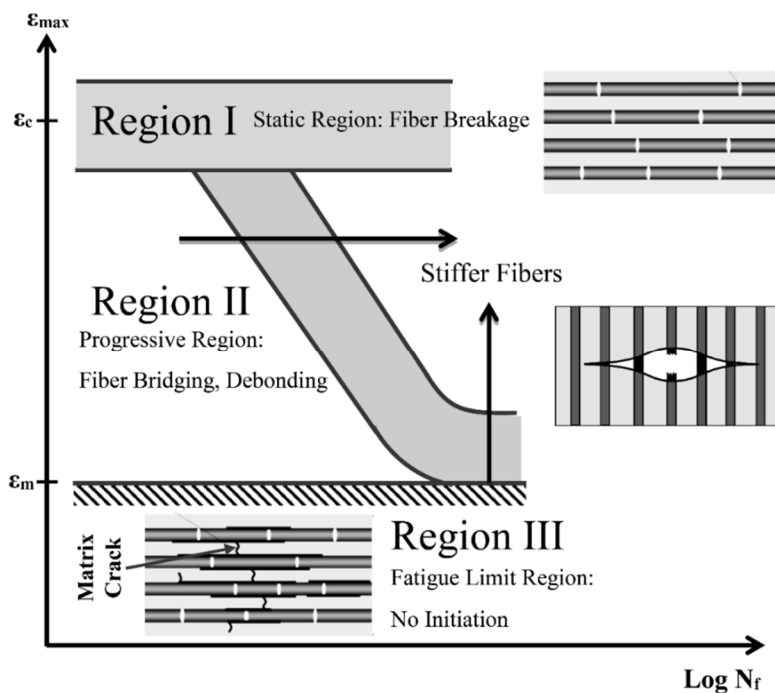


Figure 6. UD composites fatigue life diagram (FLD) (axial tension–tension loading).

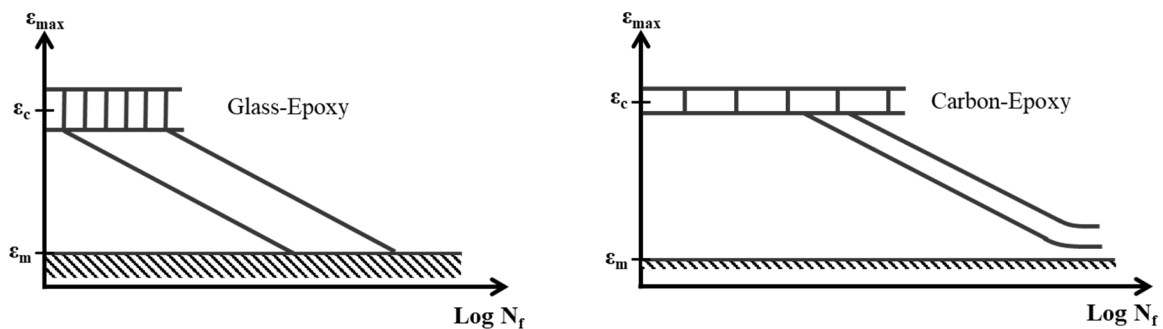
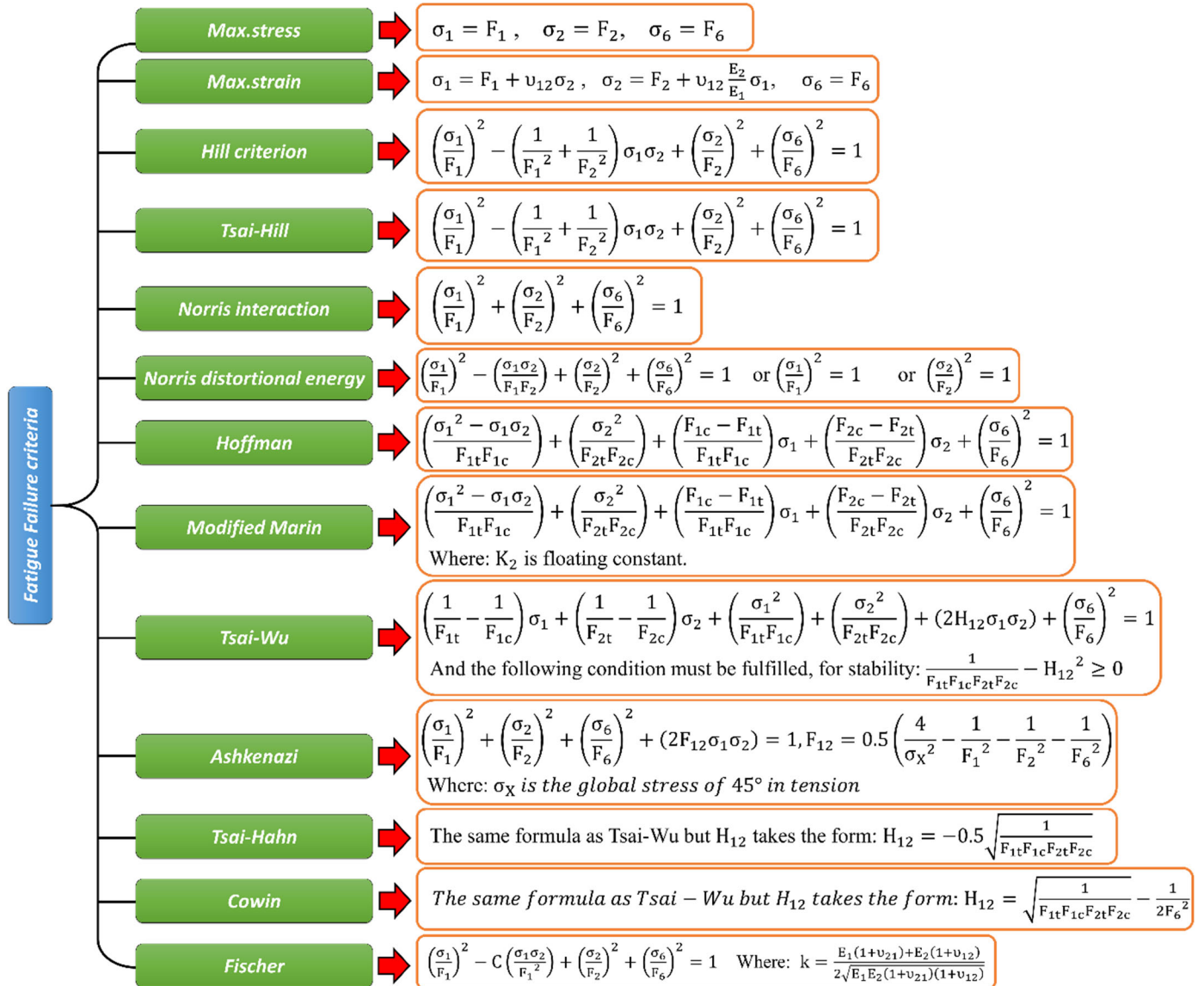


Figure 7. FLD for GFRE (left) and CFRE (right) at three fiber volume fractions [23].

#### 1.4. Fatigue failure criteria

Under fatigue loading, materials experience stress levels below their ultimate static strength; however, cyclic loading causes a progressive deterioration of material properties. Failure occurs when the residual strength degrades to the level of the

applied stress. Numerous researchers have utilized polynomial failure criteria to estimate the fatigue life of composites [45–50]. Rather than relying on static strength, these models incorporate fatigue strength as a function of cycle count within the denominator of the failure criteria, a method that significantly enhances the detection and prediction of fatigue damage. For over thirty years, global research has focused on developing such failure models to elucidate damage mechanisms. As illustrated in **Figure 8**, various theories offer different perspectives on failure, where interacting stress components collectively influence the structural integrity of the composite system. Consequently, the suitability of a specific criterion depends on the material properties and the loading conditions. In these equations, the left side represents the local stress components normalized by their respective strengths, defined as Relative Damage (R.D.), while the right side is conventionally set to unity.



**Figure 8.** Failure Criteria / Theories of Failure.

## 2. Fatigue damage detection

A standardized classification system for damage states has gained significant traction within the structural health monitoring (SHM) community. This framework characterizes damage by addressing five fundamental levels: (1) Existence: determining if damage is present; (2) Location: identifying the exact position within the structure; (3) Type: characterizing the nature of the defect; (4) Extent: assessing the severity of the damage; and (5) Prognosis: estimating the remaining useful life of the system [51].

To figure out what kind of damage a structure has and how bad it is, one usually needs to know how that structure behaves under different failure conditions. This is often done using analytical models. For instance, when monitoring the structure while it's in use, compare the modal parameters of the damaged structure to those of the structure when it's intact. This process is known as global diagnosis. If find that there's damage, can then use a model of the damaged structure to identify where the damage is located, which is referred to as local diagnosis [52].

Fatigue damage monitoring in composite materials is generally categorized into destructive testing (DT) and NDT. DT methodologies encompass laboratory and in-situ assessments designed to evaluate material limits through tensile, compression, interlaminar shear, and fracture toughness testing. Furthermore, DT characterization often involves specialized techniques such as sectioning, de-plying, and fractography to analyze internal failure mechanisms.

NDT techniques have proven quite effective for evaluating composite structures on-site, especially when it comes to checking the integrity of laminate composites. Lately, as shown in **Figure 9**, several methods have been applied to monitor these laminate structures, including: (1) Ultrasonic testing [53–59]; (2) X-Ray Radiography [60–63]; (3) Thermography [64–68]; (4) Acoustic Emission [69–73]; (5) Vibrography [74–77]; (6) Eddy Currents [78]; (7) Optical-based NDT methods like embedded Optical Fiber Sensors [79–82]; and (8) Lamb Waves techniques [83–93].

There are quite a few different NDE techniques available for monitoring fatigue damage in composite structures, each with its own strengths and weaknesses. While each method works best in its own area, they may overlap when it comes to the size of the defects that can be found, how accurately they can be detected, and which types of damage they can pick up. For instance, ultrasonics are good at finding delaminations, but acoustic emission is capable of detecting both delaminations and fiber failures. Yet, it's important to note that no single technique can identify every type of fatigue damage out there [94]. **Figure 10** presents the different types of fatigue damage that these NDE techniques can detect in laminate composite structures.

As illustrated in **Figure 10**, no single NDT method is capable of fully capturing the diverse fatigue damage states inherent in laminate composites. A multi-modal NDE approach is essential to obtain a comprehensive assessment of damage types and their impact on residual mechanical properties. For instance, integrating data from ultrasonic testing and X-ray radiography can facilitate a three-dimensional reconstruction of the intricate damage networks within the composite.

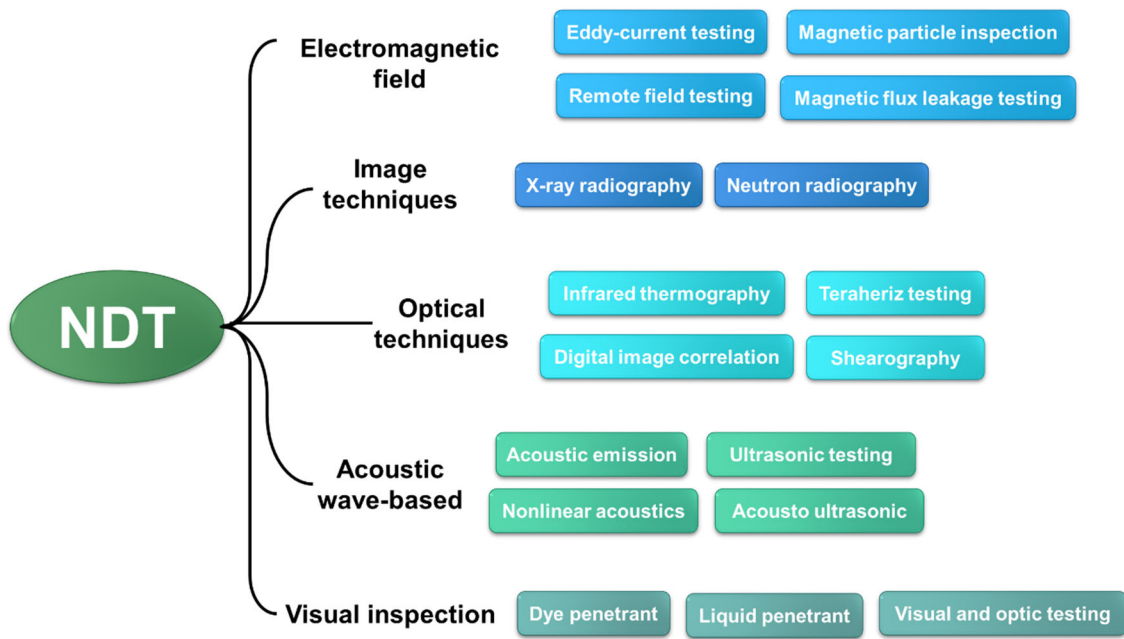


Figure 9. NDT Types for laminated composite structures.

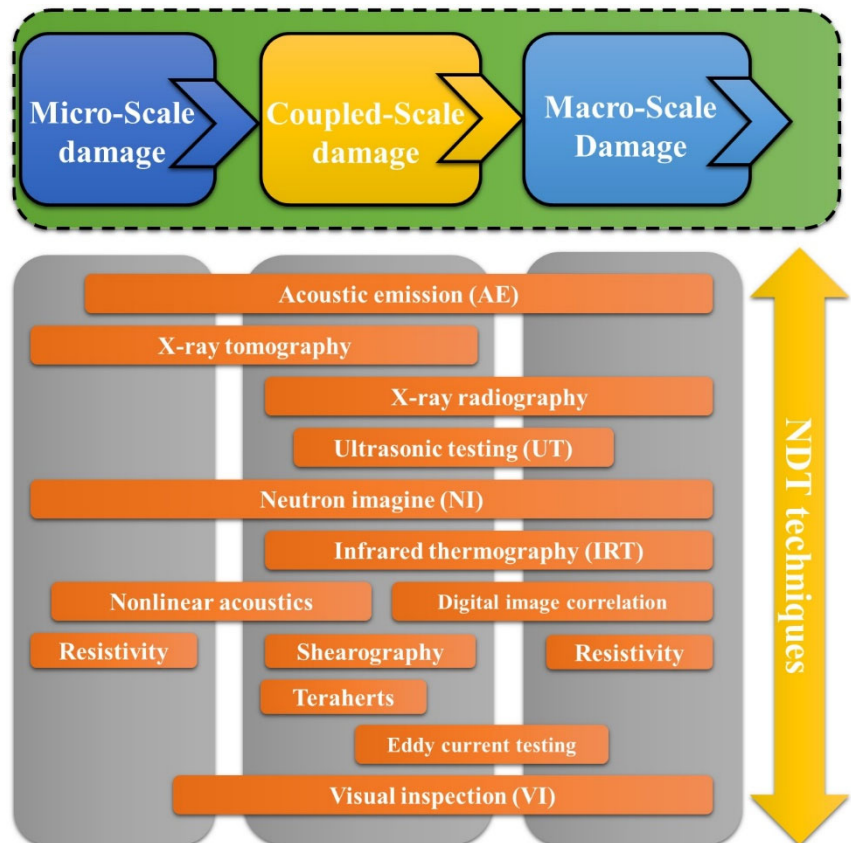


Figure 10. Overview of NDT Methods capabilities for SHM in laminated composite structures.

### 3. Damage detection-based lamb waves technique

LWs are a class of GWs that propagate through plate or shell-like structures. Over the past two decades, significant research and development have been driven by an

increasing interest in utilizing LWs for structural damage monitoring. The technology is approaching maturity for diverse engineering applications, offering a cost-effective solution for continuous and automated structural health monitoring. However, in contrast to traditional NDT methods such as ultrasonic scanning and radiography, which have been refined over fifty years, LW-based detection remains an emerging field. Several technical challenges must still be addressed before it achieves widespread industrial adoption.

LWs techniques are highly effective for NDT because they travel long distances through energy-absorbent composite materials. Their sensitivity to path interference allows for rapid screening of large areas, such as composite wing skins, while the use of different LW modes enables the detection of both surface and subsurface defects. Key advantages include non-invasive testing through insulation, comprehensive cross-sectional assessment, and cost-effective operation with minimal energy requirements [95]. Consequently, LW data analysis serves as a robust tool for diagnosing and quantifying fatigue damage, facilitating reliable estimates of the remaining useful life of composite structures.

### **3.1. History and literatures**

LWs, which were first identified by Horace Lamb back in 1917, can be found in thin, plate-like materials like panels, plates, and small beams that have parallel edges [95]. Mindlin was the pioneer in creating a detailed theory for plates, and he worked alongside Schoch and Frederick during the 1950s and 60s [96]. Then, in 1961, Worlton [97] introduced the idea of using LWs for detecting damage. Together, these contributions have paved the way for the current use of LWs in NDT.

Saravanos and Birman [98] were the first to apply LWs to composite materials, showing through both analysis and experiments that it's possible to detect delamination in composite beams using these waves. Percival and Birt [99] reached similar conclusions, directing their research toward the two main modes of LWs. As well as the other types of damage in composite materials were explored, with Seale and Smith [100] looking at fatigue and thermal damage, while Tang and Henneke [101] studied how sensitive LWs propagation is to fiber fractures.

Alleyne and Cawley [102] were the first to look at how LWs interact with defects for NDT. Following that, Saravanos et al. [103] came up with a method to detect delamination in composite materials by using LWs along with embedded PZTs. They used piezoelectric patches to generate the first anti-symmetric LWs mode.

Subsequent research into LWs techniques has focused on comparative performance against alternative composite damage detection strategies. Studies have evaluated the durability, efficacy, and sensitivity of LWs, establishing their capability for precise damage localization [104–111]. While LWs are highly effective at identifying local structural defects, their primary limitation is the requirement for an active power supply to facilitate signal generation. Furthermore, high data acquisition rates are essential to maintain adequate signal resolution. Despite these challenges, LWs offer superior sensitivity and accuracy compared to traditional methods. To optimize power consumption and data storage, researchers suggest integrating LWs into a hybrid SHM system alongside passive methods, such as Frequency Response

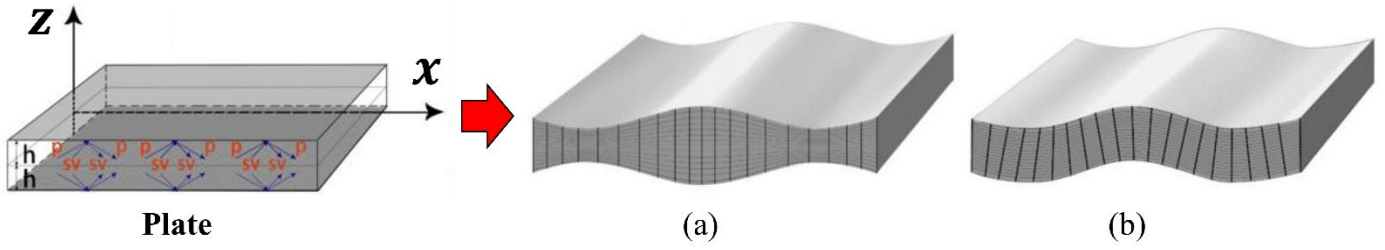
Functions (FRF) or spectral analysis. Additionally, employing wavelet transform-based analysis in the time-frequency domain has been shown to enhance the precision of damage evaluation [112–117].

Over the last decade, research has increasingly focused on overcoming the inherent challenges of using LWs for fatigue damage detection in composites. Significant efforts have been directed toward enhancing the sensitivity and localization capabilities of these techniques. For example, Yeum et al. [89] developed a novel, reference-free LW technology specifically for delamination detection. This approach enables the identification of defects along a single propagation path without requiring prior baseline data or predefined decision boundaries. By utilizing circular PZTs for wave excitation, the researchers analyzed the interaction between a single wave mode and delamination through both numerical simulations and experimental validation. Their findings demonstrate that structural integrity can be accurately assessed without the need for comparative baseline signals or multi-path data fusion.

LWs propagation can be simulated via FE analysis using platforms such as ANSYS and ABAQUS. Researchers often develop 2D cross-sectional plate models to characterize LW responses and assess fatigue-related damage, specifically debonding and delamination. Various numerical methods have been employed to model these phenomena in the time domain, including the two-dimensional Fast Fourier Transform (2D FFT) [118], wavelet transforms [119], and Semi-Analytical FE (SAFE) or Wave FE (WFE) methods [121]. FEA is frequently used to validate experimental results; for instance, Baid [122] utilized FEA to corroborate findings on skin-core interface damage in woven composite and honeycomb sandwich panels. Furthermore, Shen and Giurgiutiu [93] developed a Combined Analytical FE Approach (CAFA) for the efficient simulation of 2D LW interaction with damage. Their results, validated against scanning laser Doppler vibrometry (SLDV) data, demonstrated that CAFA offers superior computational accuracy and versatility compared to full-scale multiphysics simulations.

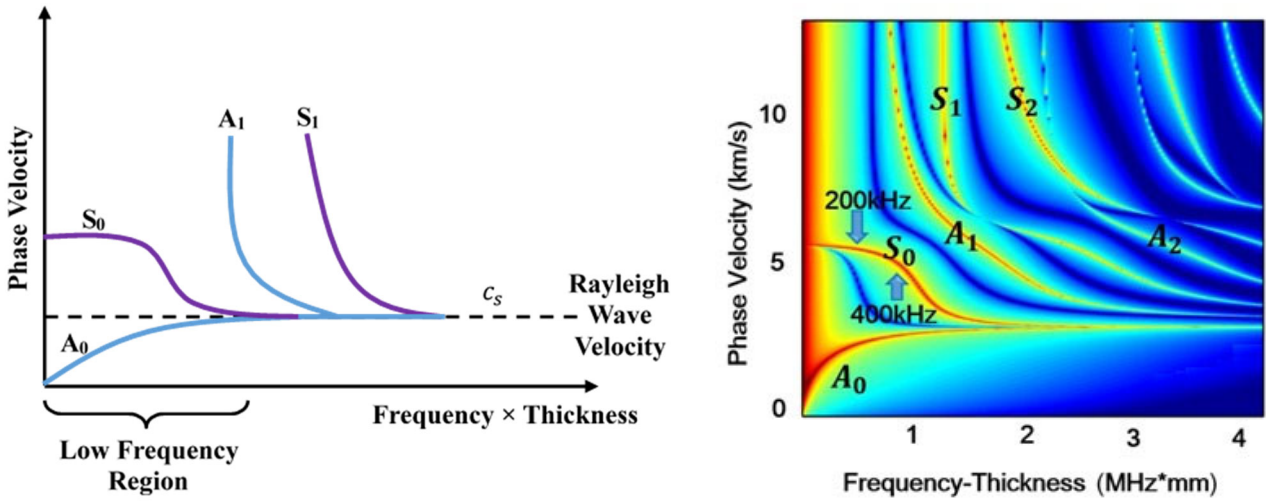
### 3.2. Modelling and simulation

LWs are ultrasonic waves that travel between two parallel surfaces, and they have been studied quite a bit for their potential in detecting damage. The materials or structures' defects can be detected by looking at the changes in phase and group velocities, along with the drop in amplitude of LWs when comparing damaged versus undamaged specimens. These waves come in two modes: symmetrical (*S*) and anti-symmetrical (*A*) modes (check out **Figure 11**). In *S* modes, the displacements of the plate are symmetrical around its center, but in some modes are anti-symmetrical. There are actually an infinite number of both *S* and *A* modes present in a plate [123].



**Figure 11.** (a) Symmetric and (b) anti-symmetric LWs modes.

LWs propagation traits are illustrated through dispersion curves, as seen in **Figure 12**. These curves plot phase and group velocities against the frequency-thickness product, which get it by solving the equations related to LWs.



**Figure 12.** LWs phase velocity dispersion curve.

The LWs dispersion curves illustrate how wavenumber relates to circular frequency  $\omega$  (or can use linear frequency  $f$  based on the equation  $\omega = 2\pi f$ ). The longitudinal wave velocity  $c_L$ , and the transverse (shear) wave velocity  $c_T$ . For isotropic plates, there's actually an analytical formula to calculate these LWs dispersion curves. These curves can be got from solving the Rayleigh-Lamb equation for a plate of thickness  $2h$ , for two distinct types of modes: Symmetric (S) and Anti-symmetric (A), [124,125] as follows:

- 1) Symmetric (S):

$$\frac{\tan(\beta h)}{\tan(\alpha h)} = - \left[ \frac{4k^2 \alpha \beta}{(k^2 - \beta^2)^2} \right] \quad (1)$$

- 2) Anti-symmetric (A):

$$\frac{\tan(\beta h)}{\tan(\alpha h)} = - \left[ \frac{(k^2 - \beta^2)^2}{4k^2 \alpha \beta} \right] \quad (2)$$

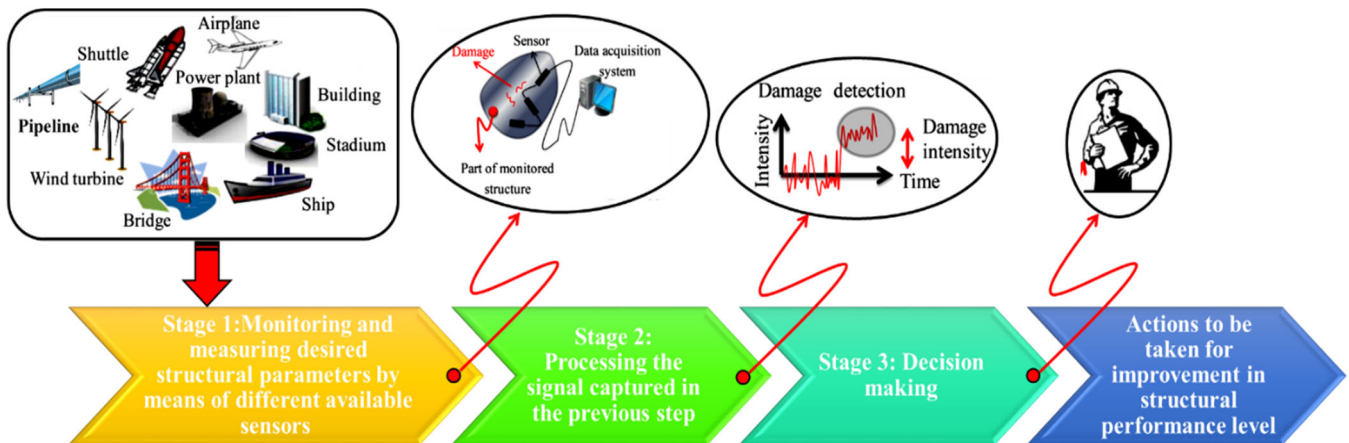
where:  $\alpha^2 = (\omega^2/c_T^2) - k^2$  and  $\beta^2 = (\omega^2/c_L^2) - k^2$ .  $c_S$  represent the vertical wave numbers for longitudinal and transverse waves, respectively.

LWs behavior in non-homogeneous composite plates tends to be trickier than in homogeneous isotropic plates, mainly because of the varying material properties. In a standard anisotropic composite plate, how the material behaves is influenced by factors like the fiber and matrix characteristics, the orientation of the fibers, and the thickness and arrangement of the layers throughout the plate.

The dispersion curves for composite plates can be calculated in basically two ways: using exact solutions or approximate ones [126]. The exact methods depend on 3D elasticity theory and are tackled through matrix techniques, like the transfer matrix method and the global matrix method [127–129]. These approaches break down the layered plates into a matrix form, capturing both stresses and displacements at the surface, along with the amplitudes of incoming and outgoing waves. On the other hand, when it comes to approximate solutions, the FE Method comes into play for figuring out dispersion curves of waveguides with various shapes. And in the case of anisotropic composite plates, the averaged material properties are actually used for each layer in the plates.

### 3.3. SHM-based lamb waves technique

SHM is an evolving field derived from the study of smart materials and structures. It has garnered significant attention recently, particularly for applications in civil infrastructure and aerospace vehicles. The primary objective of SHM is to develop automated systems capable of continuous monitoring and damage detection, thereby reducing reliance on manual inspection. As illustrated in **Figure 13**, a typical SHM architecture comprises three core components: (i) a sensory system, (ii) a data processing system, encompassing acquisition, transmission, and storage, and (iii) a health assessment system utilizing diagnostic algorithms and information management. Some advanced frameworks also incorporate a fourth component dedicated to structural repair functionalities [130–135].



**Figure 13.** SHM Strategy.

SHM utilizing LWs is categorized into passive and active systems. Passive SHM relies on sensors to detect LWs generated internally by damage events. Conversely, active SHM systems employ actuators to introduce controlled LWs into the structure, which are subsequently captured by sensors. As illustrated in **Figure 14**, these signals

provide critical data regarding structural integrity. While passive SHM is primarily utilized for damage localization, active SHM enables the selection of specific modes and frequency ranges, significantly streamlining the signal processing required for damage detection.

There are a variety of transducers that can both generate and detect LWs in composite structures. Some of these include ultrasonic transducers [136], laser transducers [137], optical fiber transducers [138], interdigital or comb transducers [139], and speckle interferometry [140]. However, PZTs are the most commonly used for employing the LWs technique to detect and locate fatigue damage in composite structures. This is mainly because they offer a strong force output at fairly low voltages, are lightweight, easy to integrate, work across a wide frequency range, are cost-effective, and respond well even at low frequencies [141–151].

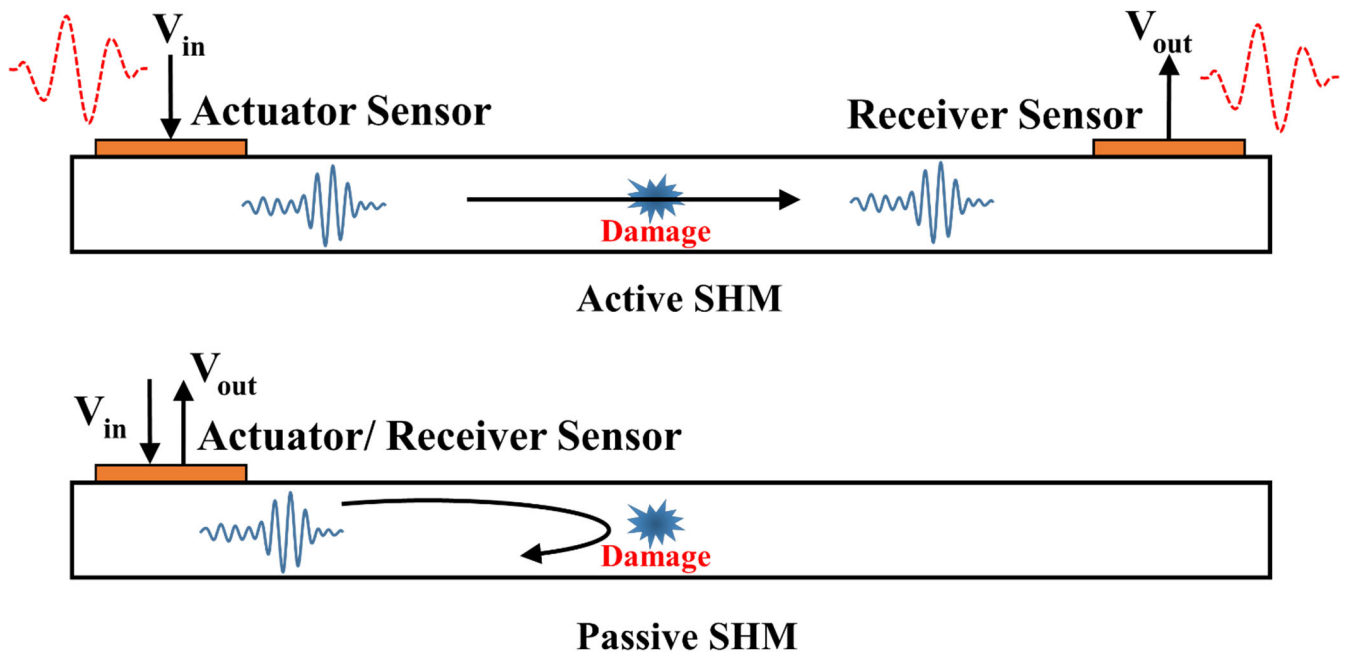


Figure 14. Active and passive SHM set up for damage detection.

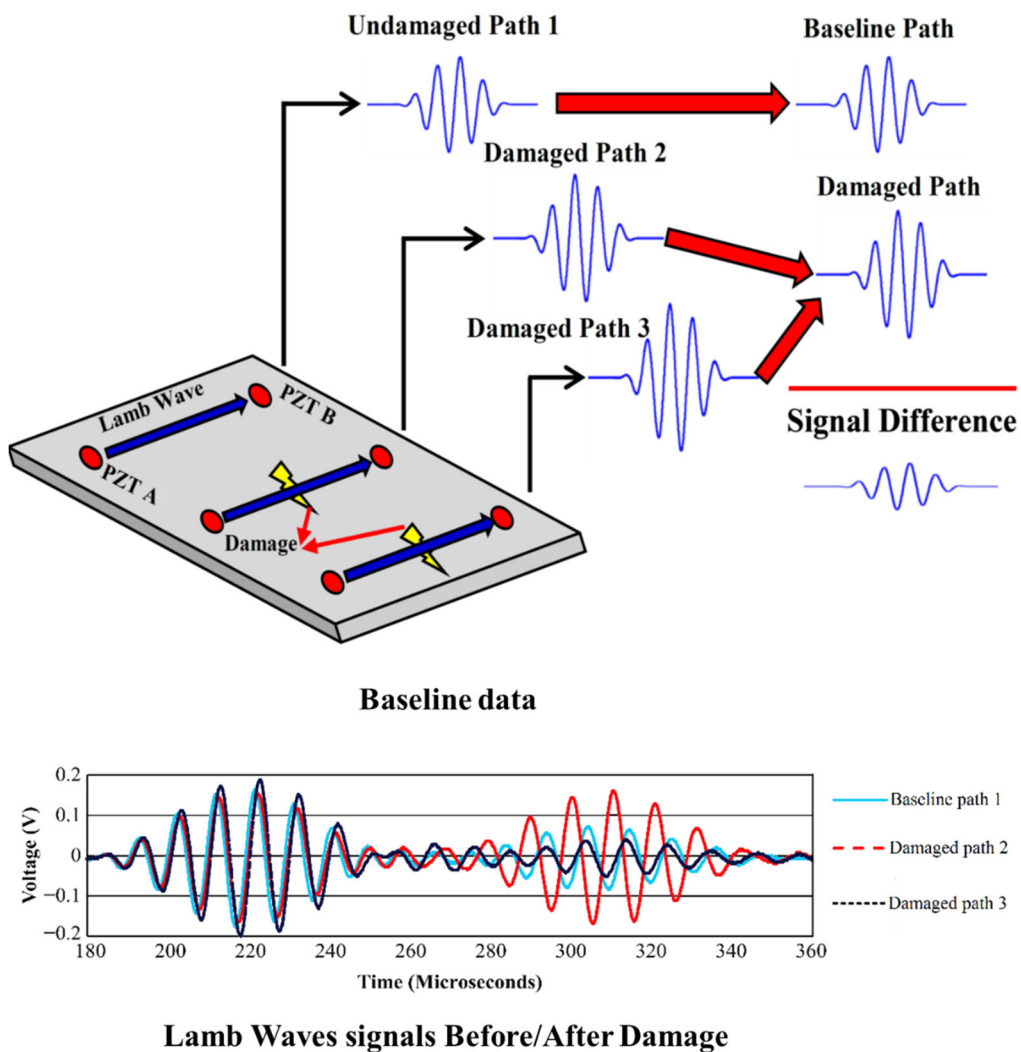
### 3.4. Damage detection process

The picked method for detecting damage using LWs needs to handle noise, vibrations from the structure, and the fact that multiple modes can overlap. Now, there are three detection methods that can be classified based on signal processing: time domain analysis [152], frequency domain analysis [153], and time-frequency domain analysis [154]. When it comes to detecting fatigue damage in composite structures with LWs techniques, the time domain correlation method is typically used for processing wave signals. With these time domain methods, the damage can be estimated by looking at the time histories of inputs, which allows for the detection of damage events both on a global scale and in specific local areas.

The damage detection model works by analyzing wave signals and comparing them to baseline measurements taken under conditions with no damage. This baseline helps remove boundary effects from getting signals when there's damage, as illustrated

in **Figure 15**. The tricky part is that it's tough to keep this baseline data consistent since it can shift due to environmental changes.

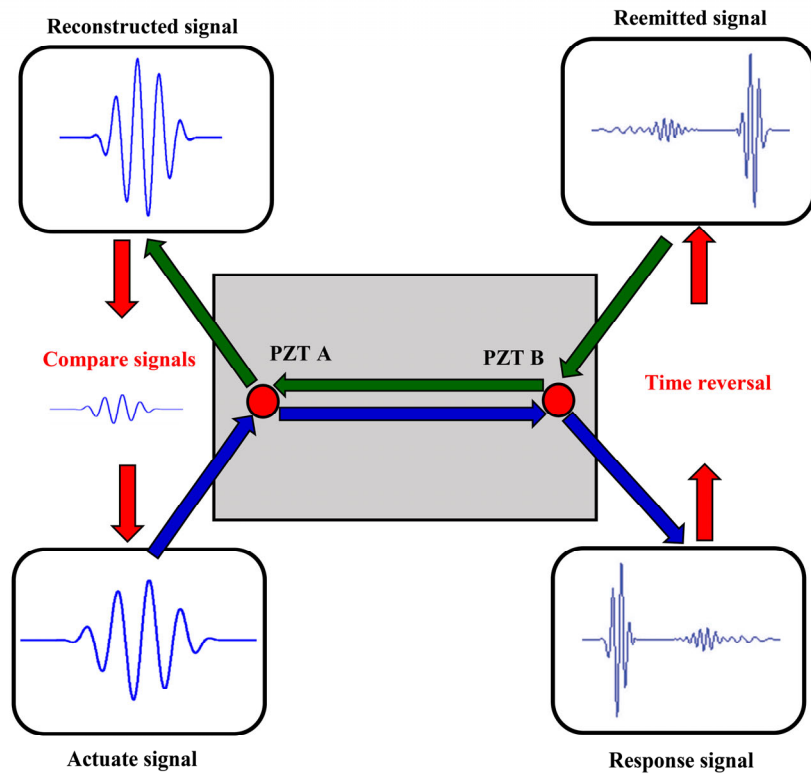
There are two detection methods that focus on the time domain: direct time and time reversal methods. In direct time domain analysis, a time series signal typically captures the propagation history of LWs moving through a structure. This gives clear insights into the waves, such as different wave modes, moving speed, attenuation and dispersion over distance, and scattering off boundaries or damages (check out **Figure 15**). The Time Reversal Method (TRM) was initially introduced to counteract the dispersion of LWs, but it has recently found a role in diagnosing damage as well. One of the big advantages of using TRM over traditional methods is that it doesn't require a baseline database [155], which can help reduce false alarms about damage because of changing environments and operational factors. So far, TRM has been effective in monitoring damage and pinpointing where it is, and it even has the potential to assess how severe and what type of damage.



**Figure 15.** The difference between the fabric Plate (Baseline) path 1 and changing the damage Path 2& Path 3.

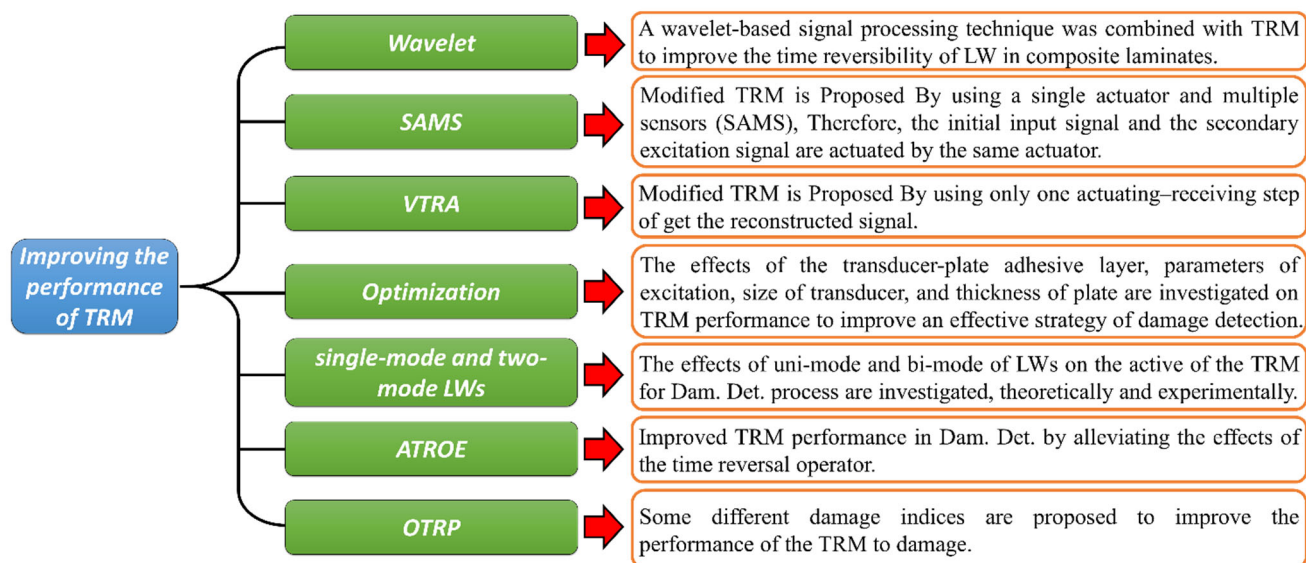
To grasp the TRM method for detecting damage, take a look at **Figure 16**. Here, a transducer array is set up on the structure, where one transducer sends out a signal

that another one picks up. Each transducer needs to act both as a sensor and an actuator, which is why PZTs are commonly used. The time reversal process shown in **Figure 16** begins with a signal being sent from PZT A and recorded by PZT B. The signal received at PZT B gets flipped in time (that is,  $V_B(t) \rightarrow V_B(-t)$ ), and it's then sent back from PZT B to PZT A, where it's recorded again. Finally, the signal at PZT A is reversed in time and compared to the original one.



**Figure 16.** Time reversal processes.

Because of the importance of the TRM, the researchers addressed improving the performance of the TRM by several methods, such as a standard reemitted signal (SRS) crossing the damage again. When the SRS is resubmitted with the damage, its effects on the SRS are compared in extensive patterns to the traditional TRM. Therefore, the SRS contains additional information on damage (such as wave velocity) that can be extracted for damage range detection. For more details of the recent studies on improving the performance of TRM, see **Figure 17**.



**Figure 17.** A comparison between recent methods for improving the performance of TRM in the LWs technique.

### 3.5. Damage localization

The damages can be localized using methods like triangulation [156], tomography techniques [157], and time reversal [158]. There's also a more complex approach that uses AI algorithms, which requires a lot of data on damage simulations to train the algorithm [159]. But if the training conditions change, the algorithm might not perform well. Additionally, the damages can be located based on how energy travels through LWs [160].

Multiple studies have addressed damage localization through various algorithmic frameworks. Rautela et al. [161] introduced a hybrid classification-regression method, highlighting the efficacy of Deep Learning (DL) in modeling guided wave behavior for damage detection. Similarly, Miorelli et al. [162] developed an automated approach for localizing and sizing damage by training a Convolutional Neural Network (CNN) on simulated signals, later validating its performance with both synthetic and experimental data. Wu et al. [163] integrated deep CNNs with continuous wavelet transforms to identify internal delamination in CFRP materials, demonstrating high precision in pinpointing defects. To improve localization efficiency using minimal sensor arrays, Liao et al. [164] proposed a multiscale CNN incorporating a self-attention module, which significantly enhanced performance on guided wave testing platforms. Furthermore, comparative studies by Feng et al. [165] on anisotropic CFRP plates indicated that ANN-based algorithms outperform probabilistic approaches in Time of Flight (ToF) analysis. Yelve et al. [166] corroborated this by utilizing Damage Index (DI) values derived from root mean square deviation, concluding that ANNs provide reliable localization. While image-based signal processing captures subtle details, it necessitates high data resolution and complicates storage requirements. Conversely, ToF and DI-based methods offer reduced computational demand but risk omitting critical signal information. Despite these advancements, achieving accurate localization across the entire structural domain—particularly at the boundaries of the sensor network—remains a significant challenge.

The time window is the period of time between  $T_{start}$  and  $T_{end}$  which are given in Equations (3) and (4), respectively:

$$T_{start} = T_1 + ToF - \frac{1}{2}T_0 \quad (3)$$

$$T_{end} = T_1 + ToF + \frac{1}{2}T_0 \quad (4)$$

where  $T_1$  denotes the wave excitation time of the actuator,  $T_0$  is the time window, ToF is the time of flight from the receiver to the actuator, and  $T_0$  is the period of the excitation wave envelope. The time window is demonstrated in **Figure 18**.

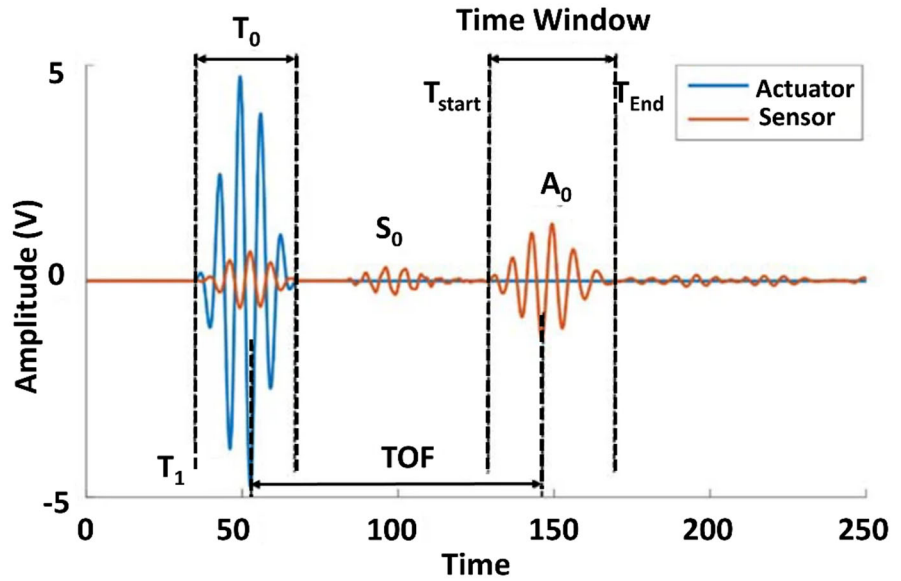


Figure 18. Failure modes on a laminate.

#### 4. Fatigue damage prediction with AI algorithms

Overall, damage detection and characterization using LWs present significant challenges. Even in simple geometries, response signals can become highly complex. Signal integrity is further complicated by fluctuations in transducer coupling conditions, which alter the amplitude of the captured data. Moreover, the generation of high-order modes at boundaries or structural discontinuities results in intricate signals that are difficult to interpret. Due to the difficulty of developing accurate physical models for LW interaction with damage in metallic and composite materials, various Artificial Intelligence (AI) algorithms, including Artificial Neural Networks (ANN), Machine Learning (ML), and Deep Learning (DL), have been proposed. These data-driven approaches facilitate structural diagnosis directly from measured signals without the need for complex physics-based modeling.

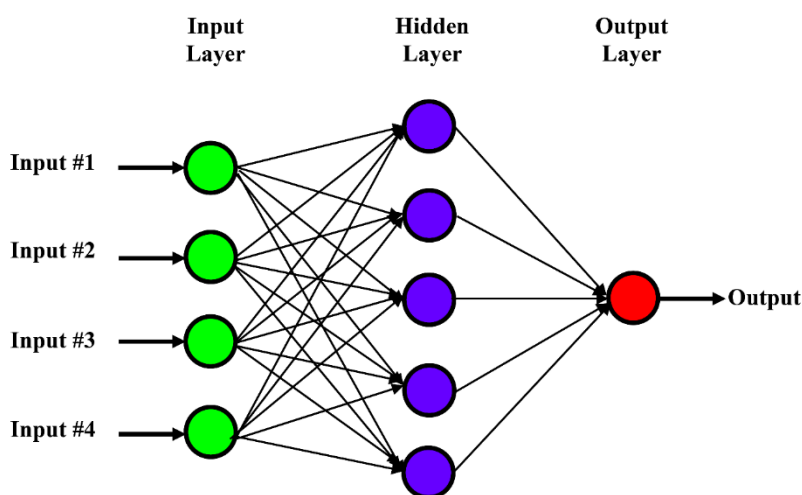
##### 4.1. Artificial neural networks (ANNs)

ANNs are a method used in AI that mimics how our brains work, sitting alongside other techniques like expert systems, genetic algorithms (GA), fuzzy logic, and Bayesian networks. It's found a lot of use in various fields such as medicine, business,

and engineering, like speech recognition, diagnosing hepatitis, fixing telecom issues from faulty software, recognizing images, and monitoring failures in laminated composite materials. The reason NNs have gained popularity is their capability to tackle non-linear problems and their resilience in noisy environments, which makes them a solid choice for these kinds of applications [167].

The main goal of ANNs is to find patterns and identify trends in complex or unclear data that are often too intricate for humans or other computer methods to catch.

ANNs come in various forms, each with its own training and testing methods. At their core, though, it usually includes numerous simple processing units called nodes or neurons [167]. These neurons affect how the others work through something called a weight. Basically, each neuron calculates a nonlinear weighted sum of its inputs and sends that result along to other neurons. The functioning of the entire network relies heavily on how these neurons interact with each other. Typically, an NN is made up of several layers of neurons: the input layer, one or more hidden layers, and the output layer, as presented in **Figure 19**. The input layer takes in the data and passes it on to the hidden layers, which handle all the complex calculations. Finally, the results are sent to the output layer, where presented to the user.

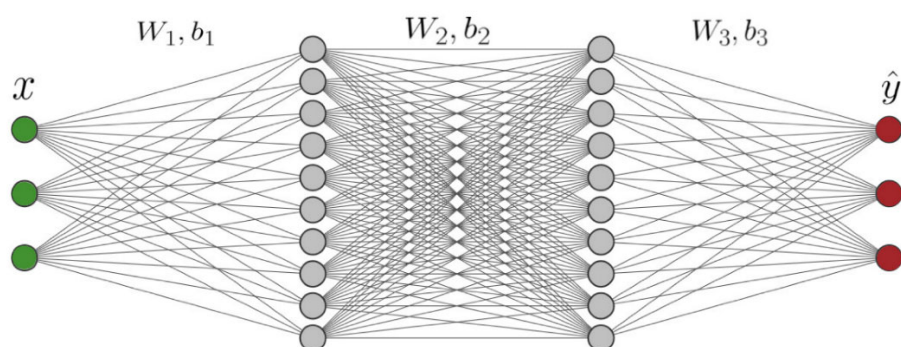


**Figure 19.** A simple ANNs.

Distinguishing between various ANNs architectures involves more than selecting a linear or nonlinear activation function; it also depends on the specific connectivity between neurons. This topology, among other design factors, defines the network structure and dictates the appropriate learning algorithms. Among the diverse architectures explored in recent research, multilayer feedforward neural networks (FFNNs) remain the most widely utilized.

These multilayer FFNNs have one or more hidden layers sitting between the input and output layers. In this setup, data moves forward from the input units to the output units without any feedback loops. This means there's no way for outputs from one layer to feed back into the inputs of the same or a previous layer. **Figure 20** below illustrates a multilayer FFNN architecture. Each layer contains a weight matrix  $W$ , a bias vector, and an output vector. To keep things organized, a layer number as a superscript is added to the variables to clarify which layer is intended.

Multilayer FFNNs possess significant computational power. Specifically, a two-layer network, utilizing a sigmoid activation function in the hidden layer and a linear function in the output layer, can approximate any functional mapping with a finite number of discontinuities. In these multilayer architectures, each layer serves a distinct functional purpose in the data transformation process.



**Figure 20.** Multilayer FFNNs.

#### 4.2. Machine learning (ML)

ML, a subset of Artificial Intelligence, enables systems to identify patterns and facilitate autonomous decision-making by learning from data rather than following explicit, task-specific programming. Through the processing of extensive datasets, these algorithms iteratively refine their performance. ML powers diverse applications, including recommendation engines, image recognition, and predictive analytics, utilizing common algorithms such as linear and logistic regression, decision trees, ANNs, K-Means clustering, K-Nearest Neighbors (KNN), and Support Vector Machines (SVMs). The standard ML training workflow involves defining the input-output architecture, acquiring vibration signals, preprocessing raw data, extracting relevant features, selecting and implementing the appropriate model, and conducting final post-processing of the results.

#### 4.3. Deep learning (DL)

DL is a more advanced branch of Machine Learning that uses multilayered ANNs to tackle tough problems, kind of like how our brains work when processing information. It is the driving force behind modern AI, including voice assistants, self-driving cars, and generative AI like large language models (LLMs). The popular DL architectures include CNNs for image processing, Recurrent Neural Networks (RNNs)/LSTM (Long Short-Term Memory) for time-series, and Generative Adversarial Networks (GANs) for generative tasks. These structures enable state-of-the-art performance in computer vision, natural language processing, and generative AI by utilizing specialized layers like convolutions, attention mechanisms, or encoder-decoder structures.

#### 4.4. Damage identification

On the side, data-driven methods use AI techniques like ML and DL to pull out complex insights from large sets of monitoring data. These techniques train classifiers

to assess damage conditions or regressors to pinpoint where the damage is located. [168–177]. Such as, Altabey et al. [178] employed a CNN to detect water absorption in BFRP composite pipes by extracting the material's dielectric features. They monitored the change in electrical capacitance corresponding to the water absorption rate from start to saturation of the BFRP. The primary CNN relates the absorption rate to the absorption time by applying the Fickian Diffusion Model (FDM); the CNN takes the potential difference between the electrodes as its input. Using the same pipe for their “case study,” Altabey [179] introduced a long-term creep thermo-mechanical fatigue behavior for CNN training, aiming to speed up the learning convergence compared to the basic steepest descent approach, and verify his network on another case study of BFRP composite plate employed by Altabey et al. [180].

Wu et al. [181] converted raw low-frequency wave signals into time-frequency scalograms via continuous wavelet transform, subsequently employing a 2D-CNN to evaluate structural integrity. In a separate study, Lee et al. [182] utilized pre-processed ultrasonic guided wave signals to train an autoencoder-based deep learning model. This architecture facilitates the extraction of damage-sensitive features from the autoencoder's latent feature space to identify fatigue in laminated composites.

Krawczuk et al. [183] investigated the integration of GA and ANNs for the detection and characterization of delamination within a numerical model of a multilayered GFRP beam. Their methodology employed two distinct approaches for identifying damage location and size: FE model updating to isolate modeling errors, and the evaluation of diverse damage scenarios. They utilized an objective function based on shifts in the first four natural frequencies, incorporating the Damage Location Assurance Criterion (DLAC) by Messina et al. [184]. While the GA successfully converged within a reasonable number of generations, the authors proposed that incorporating elitism could further optimize performance. Conversely, the ANNs demonstrated difficulty in accurately identifying the through-thickness position of the delaminated layer, a limitation attributed to an insufficient number of training cases.

Hatem et al. [185] integrated GA and ANNs to detect damage in CFRP cantilever beams. Their study categorized four distinct damage types: circular holes, delaminations, linear surface cracks, and through-thickness cracks, each varying in size, location, and orientation. To classify these defects, the researchers employed a multi-network architecture consisting of five models: a generalized regression network (GRN), a linear network, two back-propagation networks (standard and regularized), and a radial basis network. The GRN proved most effective for classification, achieving accuracy rates between 85% and 98%. Following identification, specific ANNs or GAs were utilized to quantify damage parameters. The authors concluded that while damage size could be predicted with high precision, results for location and orientation were less consistent.

Zheng et al. [186] merged computational mechanics with ANNs, particularly applying the back-propagation technique to forecast delamination in CFRP beams. They trained the ANNs on FE models that looked at various sizes and positions of delaminations, using the first five natural frequencies as their input. The team discovered that the ANNs could effectively predict both the size and the location of delamination, with just a small margin of error.

Haj-Ali et al. [187] investigated the utility of ANNs in developing nonlinear micromechanical models, identifying fiber-matrix interfacial cracking as a critical factor in damage assessment. Utilizing unit cell (UC) models for a unidirectional metal-matrix composite, they established the crack angle as a primary damage parameter. The ANN architectures were trained using data from 3D FE simulations to generate stress-strain constitutive models integrated with damage variables. This research was among the first to demonstrate the efficacy of ANNs in creating damage-inclusive micromechanical material models. Notably, however, their ANN was restricted to monotonic loading behavior and tailored specifically to a boron-aluminum metal matrix composite.

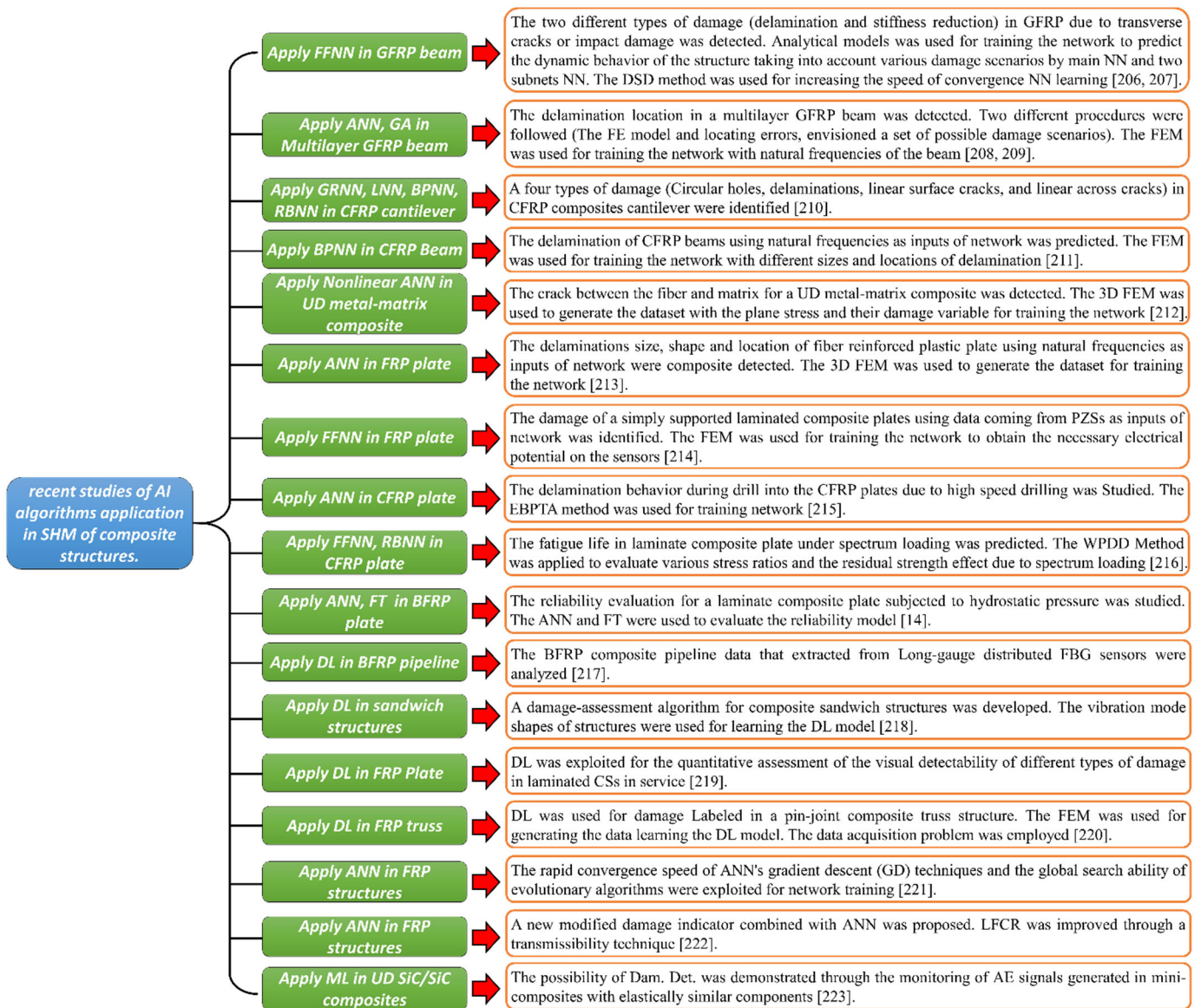
Chakraborty [188] investigated the influence of embedded delamination parameters, specifically size, geometry, and location, on the natural frequencies of FRP composite laminates. To generate a comprehensive dataset, 201 3D-FE models were developed, each featuring unique delamination configurations. Through these numerical simulations, the first ten natural frequencies were extracted for each case. This study provides a robust data-driven framework for understanding how structural modal responses are altered by internal delamination characteristics. They adjusted the numbers to fit between 0.1 and 0.9. Out of the total, they randomly picked 165 data sets to train the network, while the remaining sets were saved for testing.

Roseiro et al. [189] put forward a method that uses ANNs to detect and measure damage. They based their approach on data from PZTs, feeding this information into a FFNN. Their method employed a higher-order FE formulation, which helped capture the response of laminated composite plates and collect the necessary electrical signals from the sensors. To demonstrate how this works, they included a numerical example involving a simply supported laminated composite plate. Karnik et al. [190] looked into how delamination behaves based on different drilling parameters at the entry points of CFRP plates. For their study on high-speed drilling, they used an ANNs model that took into account factors like spindle speed, feed rate, and point angle. They went with a multilayer feed-forward ANN and trained it using the error-back-propagation training algorithm (EBPTA) to get the job done.

Roseiro et al. [189] proposed a methodology utilizing ANNs for damage detection and quantification. Their approach integrated PZT data into an FFNN, leveraging a higher-order FE formulation to accurately capture the electromechanical response of laminated composite plates. The efficacy of this method was demonstrated through a numerical example of a simply supported composite plate. In a separate study, Karnik et al. [190] investigated delamination behavior during high-speed drilling of CFRP plates. They developed a multilayer FFNN trained via the error back-propagation training algorithm (EBPTA) to analyze the influence of spindle speed, feed rate, and point angle on delamination at the entry points.

There are numerous other data-driven techniques out there, including ANN [191,192], Bayesian models [193,194], clustering algorithms [195], and various evolution algorithms [196], like extreme learning machines [197], Gaussian process regression [198], gated recurrent units [199], Kalman filters [200], random forests [201], stacked autoencoders [202], support vector machines [203], and transfer learning [204]. These methods can be categorized into shallow ML models and DL models. While shallow models can work well, their success often hinges on manual

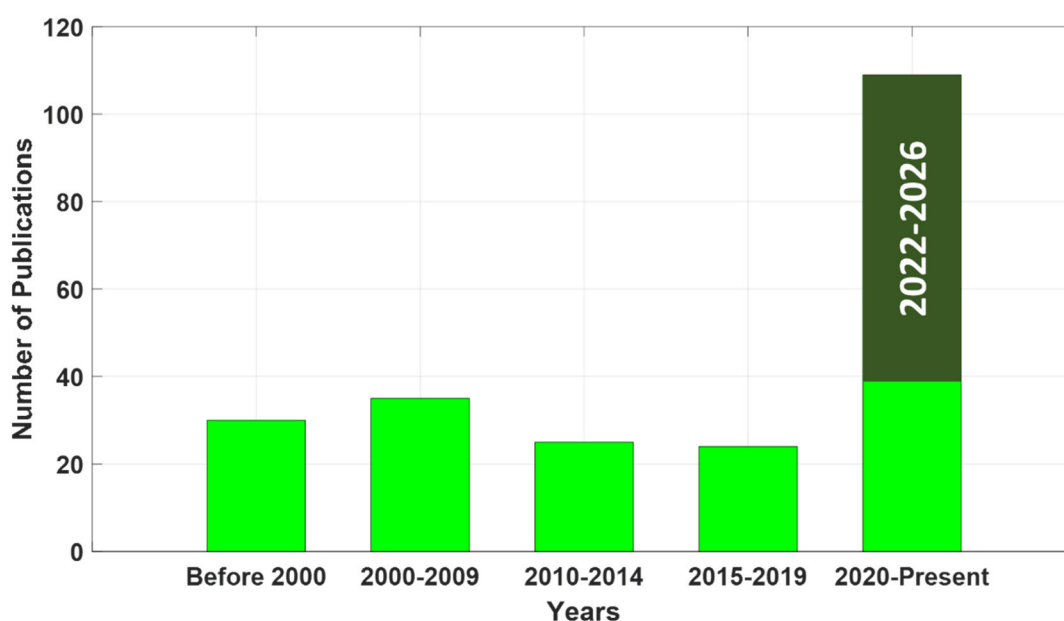
feature extraction, which is both tedious and calls for a lot of specialized knowledge [205]. Plus, different goals like damage detection, severity evaluation, and location identification each need separate feature engineering. On the other hand, DL models automatically pull out specific features needed for tasks right from raw data, making them adaptable to various objectives. So, DL-based approaches can potentially compete with, or even outdo, physics-based methods, all while needing less specialized knowledge. That said, the limited amount of monitoring data under damaged conditions might cause data-driven models to overfit, which can affect how accurately they detect and locate damage. Consequently, their reliability continues to be a point of discussion, especially in cases where data is scarce. The recent studies of AI algorithms application in SHM of composite structures are listed in **Figure 21**.



**Figure 21.** Some recent studies of AI algorithms application in SHM of composite structures [206-223].

## 5. Summary

This review looks at various aspects of SHM-based LW to identify damage in composite structures subjected to fatigue loads. First, covered different failure mechanisms, fatigue behavior, fatigue loading criteria, the advancements in SHM for composite structures, as well as SHM technologies that rely on NDT and PZT. Then, explored using the LW method for damage identification and the role of AI in SHM and damage detection. In total, we went through 223 papers, with 109 of them being published from 2020 onward. Of these, 70 papers were published between 2022 and 2023, most of which were published in 2024 and 2025, as illustrated in **Figure 22**.



**Figure 22.** Number of Publications in Google Scholar on damage identification based on guided waves in Composite structures with AI schemes, beginning from 2000 to 2026.

## 6. Discussions

In this review study, we examined LWs behavior, modeling approaches, and applications, specialized in fatigue damage detection for composite structures. This review is structured around three core components:

Part 1: In this part, we discussed the composite materials fundamentals and material overview of analyzed composite structures, failure mechanisms, and fatigue behaviors. We found that the damage progression was identified as a three-phase fatigue failure process:

- 3) Phase I: Matrix cracking initiation,
- 4) Phase II: Local delaminations developing from cracks,
- 5) Phase III: Delamination consolidation leading to final failure.

Fatigue Life Diagram (FLD) is visualized as three fatigue regions influenced by fiber, matrix, and interface properties (**Figure 6**). The Key FLD affected Factors were summarized in Loading types, frequency, volume fraction ( $V_f$ ), fiber orientation, mean

stress/stress ratio, environmental conditions, sizing, stress gradient, surface finish, and stress concentration.

Part 2: In this part, we discussed the damage monitoring techniques, especially the NDT methods and evaluated various NDT techniques with different sensitivities, such as Ultrasonic testing, X-Ray Radiography, Thermography, Acoustic Emission, Vibrography, Eddy Currents, Optical Fiber Sensors, and Lamb Waves (LWs), and introduced the detection tolerance of these techniques through **Figure 10** of the comparative analysis presented. Then we focused on the advantages of LWs technique for the damage monitoring in composite structures, which are limited to:

- 1) Large Area Coverage: Sparse transducer setup enables extensive monitoring with minimal area loss,
- 2) Cross-Section Inspection: Multiple wave modes facilitate internal and surface defect monitoring,
- 3) Damage Classification: Wave mode variations enable differentiation of damage types,
- 4) High Sensitivity: Superior detection accuracy rates,
- 5) Accessibility: Effective for enclosed/insulated structures (e.g., underwater pipelines),
- 6) Integration Potential: Compatible with structural health monitoring (SHM) systems for automated online detection,
- 7) Efficiency Benefits: Reduced energy consumption and operational costs (requires advanced signal processing),

In summary, we found LWs demonstrate significant potential for composite damage identification. Method adaptability allows customization for various damage scenarios. Presents a viable alternative to traditional NDE approaches. The revised format maintains all technical content while improving readability through logical grouping and consistent bullet-point structure.

Part 3: In this part, we discussed the integration with AI for enhancing the composite damage identification. We found that combining LWs with AI algorithms (ANNs, ML, DL) creates an optimized damage identification system, where as:

- 1) AI models trained on LWs data can effectively predict damage location and severity with minimal error,
- 2) Simplifies complex signal relationships while saving processing time and resources,
- 3) Enables power-efficient, real-time monitoring with reduced data storage requirements,
- 4) Requires simulated damage data for effective training of AI models.

Overall Recommendation: The study concludes that Lamb Wave technology, when integrated with AI-driven SHM systems, provides a highly accurate and efficient solution for fatigue damage detection in composite structures, offering significant advantages in sensitivity, coverage, and operational efficiency.

## **7. Conclusion**

In the end, Fatigue in composites progresses through three phases—matrix cracking, delamination, and final failure—governed by material properties and loading

conditions. LWs techniques offer superior structural health monitoring, providing large-area, automated, and deep-penetration damage detection compared to traditional methods. Integrating LW data with AI algorithms (ANNs, ML, DL) enables accurate damage localization while optimizing computational efficiency.

The study concludes that LWs technology, when integrated with AI-driven SHM systems, provides a highly accurate and efficient solution for fatigue damage detection in composite structures, offering significant advantages in sensitivity, coverage, and operational efficiency.

## 8. Future outlook

The future outlook for AI-integrated guided wave damage identification in composite structures focuses on transitioning from simple detection to autonomous, real-time diagnostic and prognostic systems. Current trends emphasize moving beyond localization to full damage characterization (size, shape, and type) and the prediction of Remaining Useful Life (RUL).

**Conflict of interest:** The author declare no conflicts of interest to report regarding the present study.

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